

## State Feedback and Observers

### Real-Time Systems, Lecture 9

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## Lecture 9

[IFAC PB Chapter 8]

- State feedback
- Observers
- Integral action and disturbance estimation

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## Control Design

Many factors to consider, including:

- Attenuation of load disturbances
- Reduction of the effect of measurement noise
- Command signal following
- Variations and uncertainties in process behavior

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## Two Classes of Control Problems

**Regulation problems:** compromise between rejection of load disturbances and injection of measurement noise

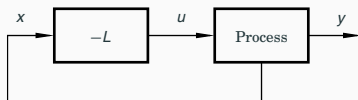
- Feedback
- Lecture 9

**Servo problems:** make the output respond to command signals in the desired way

- Feedforward
- Lecture 10

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## State Feedback: Problem Formulation



- Discrete-time process model

$$x(k+1) = \Phi x(k) + \Gamma u(k)$$

- Linear feedback from all state variables

$$u(k) = -Lx(k)$$

- Disturbances modelled by nonzero initial state  $x(0) = x_0$
- Goal: Control the state to the origin, using a reasonable control signal

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## Closed-Loop System

The state equation

$$x(k+1) = \Phi x(k) + \Gamma u(k)$$

with the control law

$$u(k) = -Lx(k)$$

gives the closed-loop system

$$x(k+1) = (\Phi - \Gamma L)x(k)$$

Pole placement design: Choose  $L$  to obtain the desired characteristic equation

$$\det(zI - \Phi + \Gamma L) = 0$$

(Matlab: `place` or `acker`)

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### Example – Double Integrator

$$x(k+1) = \begin{pmatrix} 1 & h \\ 0 & 1 \end{pmatrix} x(k) + \begin{pmatrix} h^2/2 \\ h \end{pmatrix} u(k)$$

Linear state-feedback controller

$$u(k) = -Lx(k) = -l_1x_1(k) - l_2x_2(k)$$

The closed-loop system becomes

$$x(k+1) = (\Phi - \Gamma L)x(k) = \begin{pmatrix} 1 - l_1h^2/2 & h - l_2h^2/2 \\ -l_1h & 1 - l_2h \end{pmatrix} x(k)$$

Characteristic equation

$$z^2 + \left(\frac{l_1h^2}{2} + l_2h - 2\right)z + \left(\frac{l_1h^2}{2} - l_2h + 1\right) = 0$$

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### Example Cont'd

Characteristic equation

$$z^2 + \left(\frac{l_1h^2}{2} + l_2h - 2\right)z + \left(\frac{l_1h^2}{2} - l_2h + 1\right) = 0$$

Assume desired characteristic equation  $z^2 + a_1z + a_2 = 0$ .

Linear equations for  $l_1$  and  $l_2$

$$\frac{l_1h^2}{2} + l_2h - 2 = a_1 \quad \frac{l_1h^2}{2} - l_2h + 1 = a_2$$

Solution:

$$l_1 = \frac{1}{h^2}(1 + a_1 + a_2)$$

$$l_2 = \frac{1}{2h}(3 + a_1 - a_2)$$

- $L$  depends on  $h$

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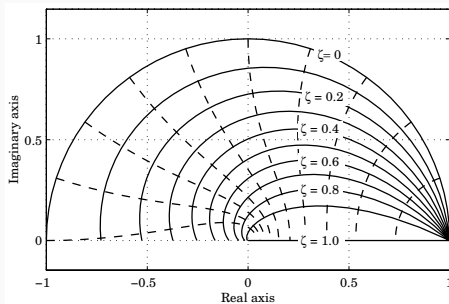
### Where to Place the Poles?

Recall from Lecture 7:

Loci of constant  $\zeta$  (solid) and  $\omega h$  (dashed) when

$$\frac{\omega^2}{s^2 + 2\zeta\omega s + \omega^2}$$

is sampled using ZOH ( $z_i = e^{s_i h}$ )

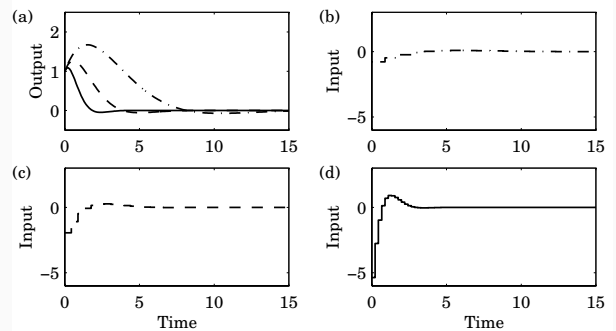


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### Example – Choice of Design Parameters

Double integrator,  $x_0^T = [1 \ 1]$ ,  $\omega h = 0.44$ ,  $\zeta = 0.707$

(b)  $\omega = 0.5$  (dash-dotted), (c)  $\omega = 1$  (dashed), (d)  $\omega = 2$  (solid)



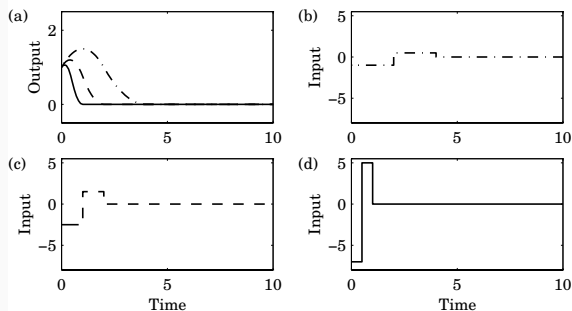
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### Deadbeat Control – Only in Discrete Time

Choose  $P(z) = z^n \Rightarrow h$  only remaining design parameter

Drives all states to zero in at most  $n$  steps after an impulse disturbance in the states (can be very aggressive for small  $h$ !)

Example: Double integrator,  $x_0^T = [1 \ 1]$



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### Controllability

The eigenvalues of  $\Phi - \Gamma L$  can be placed arbitrarily if and only if the system is *controllable*, i.e. if the controllability matrix

$$W_c = \begin{bmatrix} \Gamma & \Phi\Gamma & \dots & \Phi^{n-1}\Gamma \end{bmatrix}$$

has full rank.

In practice, moving some eigenvalues could require high gain and lead to bad controllers.

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## State Feedback in Controllable Form

Convert the system to controllable canonical form:

$$x(k+1) = \begin{bmatrix} -a_1 & -a_2 & \dots & -a_n \\ 1 & 0 & \dots & 0 \\ & \ddots & & \vdots \\ & & 1 & 0 \end{bmatrix} x(k) + \begin{bmatrix} 1 \\ 0 \\ \vdots \\ 0 \end{bmatrix} u(k)$$

In this case, application of the state feedback

$$u = -l_1 x_1 - \dots - l_n x_n$$

changes the coefficients  $a_1, \dots, a_n$  to  $a_1 + l_1, \dots, a_n + l_n$ , so the characteristic polynomial changes to

$$z^n + (a_1 + l_1)z^{n-1} + \dots + (a_{n-1} + l_{n-1})z + a_n + l_n$$

Design method: Transform to controllable canonical form, apply state feedback, transform the controller back again – Ackermann's formula (see IFAC PB)

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## State Feedback with Integral Action

Integral action can be introduced by augmenting the plant model with an extra state variable,  $x_i$ , that integrates the plant output:

$$x_i(k+1) = x_i(k) + y(k) = x_i(k) + Cx(k)$$

The augmented open-loop system becomes

$$\begin{bmatrix} x(k+1) \\ x_i(k+1) \end{bmatrix} = \begin{bmatrix} \Phi & 0 \\ C & I \end{bmatrix} \begin{bmatrix} x(k) \\ x_i(k) \end{bmatrix} + \begin{bmatrix} \Gamma \\ 0 \end{bmatrix} u(k)$$

We can then design a state feedback controller

$$u(k) = - \begin{bmatrix} L & L_i \end{bmatrix} \begin{bmatrix} x(k) \\ x_i(k) \end{bmatrix}$$

using the same techniques as before

(Integral action can also be introduced using a disturbance observer, as we will see later)

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## Reconstruction

What to do if we cannot measure the full state vector or if we have noisy measurements?

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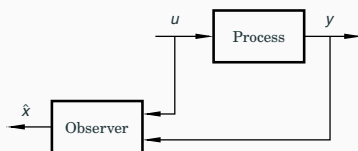
## Reconstruction Through Direct Calculations

Basic idea: Reconstruct the state vector  $x(k)$  through direct calculations using the output and input sequences  $y(k), y(k-1), \dots, u(k-1), u(k-2), \dots$  together with the state-space model of the plant.

See IFAB PB p. 57 for details.

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## Reconstruction Using An Observer



Simulated process model:

$$\begin{aligned} \hat{x}(k+1) &= \Phi \hat{x}(k) + \Gamma u(k) \\ \hat{y}(k) &= C \hat{x}(k) \end{aligned}$$

Introduce "feedback" from measured  $y(k)$

$$\hat{x}(k+1) = \Phi \hat{x}(k) + \Gamma u(k) + K(y(k) - C \hat{x}(k))$$

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## Reconstruction Using An Observer

Form the estimation error  $\tilde{x} = x - \hat{x}$

$$\begin{aligned} \tilde{x}(k+1) &= \Phi \tilde{x}(k) - KC \tilde{x}(k) \\ &= [\Phi - KC] \tilde{x}(k) \end{aligned}$$

- Any observer pole placement possible, provided the observability matrix

$$W_o = \begin{bmatrix} C \\ \vdots \\ C\Phi^{n-1} \end{bmatrix}$$

has full rank

- Choose  $K$  to get good convergence but not too much amplification of measurement noise

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## Deadbeat Observer

A *deadbeat observer* is obtained if the observer gain  $K$  is chosen so that the matrix  $\Phi - KC$  has all eigenvalues zero.

The observer error goes to zero in finite time (in at most  $n$  steps, where  $n$  is the order of the system)

Noise sensitive (fast observer dynamics)

Equivalent to reconstruction using direct calculations.

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## Observer for the Double Integrator

$$\Phi - KC = \begin{pmatrix} 1 & h \\ 0 & 1 \end{pmatrix} - \begin{pmatrix} k_1 \\ k_2 \end{pmatrix} \begin{pmatrix} 1 & 0 \end{pmatrix} = \begin{pmatrix} 1 - k_1 & h \\ -k_2 & 1 \end{pmatrix}$$

Characteristic equation

$$z^2 - (2 - k_1)z + 1 - k_1 + k_2h = 0$$

Desired characteristic equation:

$$z^2 + p_1z + p_2 = 0$$

Gives:

$$\begin{aligned} 2 - k_1 &= -p_1 \\ 1 - k_1 + k_2h &= p_2 \end{aligned}$$

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## Observer for the Double Integrator cont'd

Solution:

$$\begin{aligned} k_1 &= 2 + p_1 \\ k_2 &= (1 + p_1 + p_2)/h \end{aligned}$$

Assume deadbeat observer ( $p_1 = p_2 = 0$ )

$$\begin{aligned} k_1 &= 2 \\ k_2 &= 1/h \end{aligned}$$

Resulting observer (assuming  $u = 0$ )

$$\begin{aligned} \hat{x}_1(k+1) &= \hat{x}_1(k) + h\hat{x}_2(k) + 2(y(k) - \hat{x}_1(k)) \\ \hat{x}_2(k+1) &= \hat{x}_2(k) + \frac{1}{h}(y(k) - \hat{x}_1(k)) \end{aligned}$$

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## An Alternative Observer

The observer presented so far has a one sample delay:

$\hat{x}(k | k-1)$  depends only on measurements up to time  $k-1$

Alternative observer with direct term:

$$\begin{aligned} \hat{x}(k | k) &= \Phi \hat{x}(k-1 | k-1) + \Gamma u(k-1) \\ &\quad + K[y(k) - C(\Phi \hat{x}(k-1 | k-1) + \Gamma u(k-1))] \\ &= (I - KC)(\Phi \hat{x}(k-1 | k-1) + \Gamma u(k-1)) + Ky(k) \end{aligned}$$

Reconstruction error:

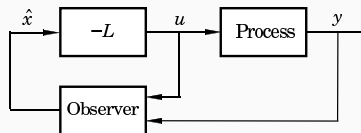
$$\tilde{x}(k | k) = x(k) - \hat{x}(k | k) = (\Phi - KC\Phi)\tilde{x}(k-1 | k-1)$$

- $\Phi - KC\Phi$  can be given arbitrary eigenvalues if  $\Phi - KC$  can
- $K$  may be chosen so that some of the states will be observed directly through  $y \Rightarrow$  the order of the observer can be reduced
  - Reduced order observer or *Luenberger observer*

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## Output Feedback

State feedback from observed state:



Controller:

$$\begin{aligned} \hat{x}(k+1) &= \Phi \hat{x}(k) + \Gamma u(k) + K(y(k) - C\hat{x}(k)) \\ u(k) &= -L\hat{x}(k) \end{aligned}$$

Controller transfer function (from  $y$  to  $u$ ):

$$H_c(z) = -L(zI - \Phi + \Gamma L + KC)^{-1}K$$

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## Analysis of the Closed-Loop System

$$\begin{aligned} x(k+1) &= \Phi x(k) + \Gamma u(k) \\ \tilde{x}(k+1) &= (\Phi - KC)\tilde{x}(k) \\ u(k) &= -L\hat{x}(k) = -L(x(k) - \tilde{x}(k)) \end{aligned}$$

Eliminate  $u(k)$

$$\begin{pmatrix} x(k+1) \\ \tilde{x}(k+1) \end{pmatrix} = \begin{pmatrix} \Phi - \Gamma L & \Gamma L \\ 0 & \Phi - KC \end{pmatrix} \begin{pmatrix} x(k) \\ \tilde{x}(k) \end{pmatrix}$$

Separation

$$\begin{aligned} \text{Control poles: } &A_c(z) = \det(zI - \Phi + \Gamma L) \\ \text{Observer poles: } &A_o(z) = \det(zI - \Phi + KC) \end{aligned}$$

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## Disturbance Estimation

How to handle disturbances that can not be modeled as impulse disturbances in the process state?

Assume that the process is described by

$$\begin{aligned}\frac{dx}{dt} &= Ax + Bu + v \\ y &= Cx\end{aligned}$$

where  $v$  is a disturbance modeled as

$$\begin{aligned}\frac{dw}{dt} &= A_w w \\ v &= C_w w\end{aligned}$$

Since disturbances typically have most of their energy at low frequencies,  $A_w$  often has eigenvalues in the origin (constant disturbance) or on the imaginary axis (sinusoidal disturbance)

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## Disturbance Estimation

Augment the state vector:  $\begin{pmatrix} x \\ w \end{pmatrix}$

Gives the augmented system

$$\begin{aligned}\frac{d}{dt} \begin{pmatrix} x \\ w \end{pmatrix} &= \begin{pmatrix} A & C_w \\ 0 & A_w \end{pmatrix} \begin{pmatrix} x \\ w \end{pmatrix} + \begin{pmatrix} B \\ 0 \end{pmatrix} u \\ y &= \begin{pmatrix} C & 0 \end{pmatrix} \begin{pmatrix} x \\ w \end{pmatrix}\end{aligned}$$

Sample this using ZOH:

$$\begin{aligned}\begin{pmatrix} x(k+1) \\ w(k+1) \end{pmatrix} &= \begin{pmatrix} \Phi & \Phi_{xw} \\ 0 & \Phi_w \end{pmatrix} \begin{pmatrix} x(k) \\ w(k) \end{pmatrix} + \begin{pmatrix} \Gamma \\ 0 \end{pmatrix} u(k) \\ y &= \begin{pmatrix} C & 0 \end{pmatrix} \begin{pmatrix} x(k) \\ w(k) \end{pmatrix}\end{aligned}$$

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## Augmented Observer and State Feedback

Augmented observer:

$$\begin{pmatrix} \hat{x}(k+1) \\ \hat{w}(k+1) \end{pmatrix} = \begin{pmatrix} \Phi & \Phi_{xw} \\ 0 & \Phi_w \end{pmatrix} \begin{pmatrix} \hat{x}(k) \\ \hat{w}(k) \end{pmatrix} + \begin{pmatrix} \Gamma \\ 0 \end{pmatrix} u(k) + \begin{pmatrix} K \\ K_w \end{pmatrix} \epsilon(k)$$

with  $\epsilon(k) = y(k) - C\hat{x}(k)$

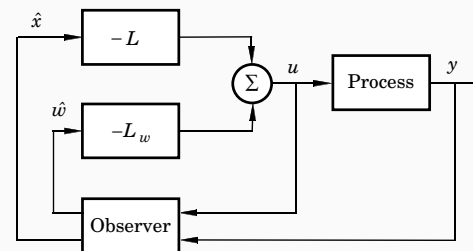
Augmented state feedback control law:

$$u(k) = -L\hat{x}(k) - L_w\hat{w}(k)$$

If possible, select  $L_w$  such that  $\Phi_{xw} - \Gamma L_w = 0$

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## Disturbance Estimation: Block Diagram



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## Disturbance Estimation: Closed-Loop System

The closed-loop system can be written

$$\begin{aligned}x(k+1) &= (\Phi - \Gamma L)x(k) + (\Phi_{xw} - \Gamma L_w)w(k) + \Gamma L\tilde{x}(k) + \Gamma L_w\tilde{w}(k) \\ w(k+1) &= \Phi_w w(k) \\ \tilde{x}(k+1) &= (\Phi - KC)\tilde{x}(k) + \Phi_{xw}\tilde{w}(k) \\ \tilde{w}(k+1) &= \Phi_w\tilde{w}(k) - K_w C\tilde{x}(k)\end{aligned}$$

- $L$  ensures that  $x$  goes to zero at the desired rate after a disturbance.
- The gain  $L_w$  reduces the effect of the disturbance  $v$  on the system by feedforward from the estimated disturbance  $\hat{w}$ .
- $K$  and  $K_w$  influence the rate at which the estimation errors go to zero.

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## Special Case: Constant Input Disturbance

Assume constant disturbance acting on the plant input:

- $v = w$
- $\Phi_w = 1$
- $\Phi_{xw} = \Gamma$

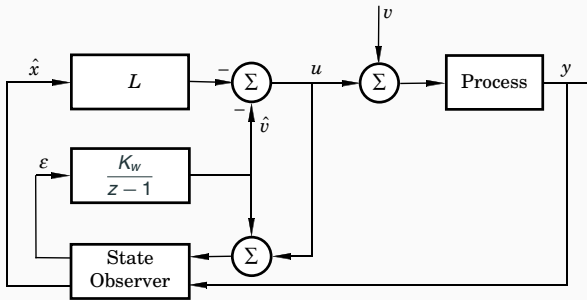
If we choose  $L_w = 1$  we will have perfect cancellation of the load disturbance

New controller + estimator

$$\begin{aligned}u(k) &= -L\hat{x}(k) - \hat{v}(k) \\ \hat{x}(k+1) &= \Phi\hat{x}(k) + \Gamma(\hat{v}(k) + u(k)) + K\epsilon(k) \\ \hat{v}(k+1) &= \hat{v}(k) + K_w\epsilon(k) \quad (\text{integrator}) \\ \epsilon(k) &= y(k) - C\hat{x}(k)\end{aligned}$$

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## Special Case: Block Diagram



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## Example – Design

- Control of double integrator

$$\frac{dx}{dt} = \begin{pmatrix} 0 & 1 \\ 0 & 0 \end{pmatrix} x + \begin{pmatrix} 0 \\ 1 \end{pmatrix} u$$

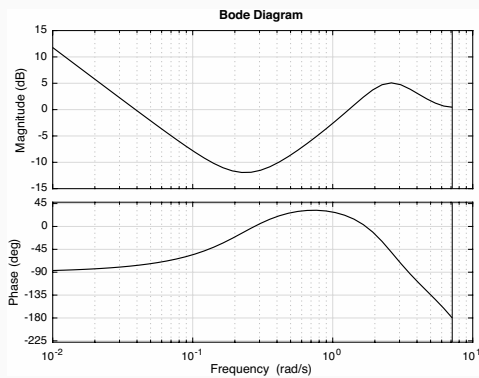
$$y = \begin{pmatrix} 1 & 0 \end{pmatrix} x$$

- Sample with  $h = 0.44$
- Discrete state feedback designed based on continuous-time specification  $\omega = 1$ ,  $\zeta = 0.7$ 
  - Gives  $L = [0.73 \ 1.21]$
- Extended observer assuming constant input disturbance to obtain integral action; all three poles placed in  $z = 0.75$ .

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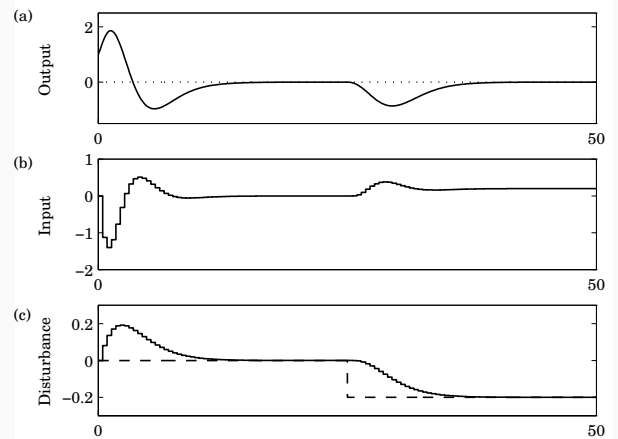
## Example – Design

Bode diagram of resulting controller:



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## Example – Simulation



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## Optimization-Based Design

Pole placement design used in this course:

- $L$  and  $K$  derived through pole placement

In the course Multivariable Control (Flervariabel Reglering),  $L$  and  $K$  are instead derived through optimization

- LQ (Linear Quadratic) and LQG (Linear Quadratic Gaussian) control
- Short overview in Chapter 11 of IFAC PB
- Not part of this course

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## Example in Matlab

```
>> A = [0 1; 0 0];
>> B = [0; 1];
>> h = 0.44;
>> [Phi, Gamma] = c2d(A, B, h)
>> Hp = ss(Phi, Gamma, C, 0, h);
>> % Desired poles in continuous time
>> omega = 1; zeta = 0.7;
>> pc = roots([1 2*zeta*omega omega^2])
>> % Corresponding desired discrete-time poles
>> pd = exp(pc*h)
>> % Design state feedback
>> L = place(Phi, Gamma, pd)
>> Le = [L 1];
>> % Design augmented observer
>> Phie = [Phi Gamma; zeros(1,2) 1];
>> Ce = [C 0];
>> Ke = acker(Phie', Ce', [0.75 0.75 0.75])'
>> % Form controller
>> Hc = ss(Phie-Gamma*Le-Ke*Ce, Ke, Le, 0, h);
>> bode(Hc)
```

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